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**PRIME MINITRACK AND BAKER-NUNN  
ORBITS OF SATELLITE 1959 $\alpha_1$   
(VANGUARD II)**

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NATIONAL AERONAUTICS AND SPACE ADMINISTRATION

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ABSTRACT

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Concurrent data necessary for making a comparison study of the accuracies of Prime Minitrack and Baker-Nunn observations of Satellite 1959 $\alpha_1$  (Vanguard II) are presented in this report. In all, 244 Prime Minitrack and 187 Baker-Nunn observations are available which were made over a 26-day period while the satellite's transmitter was operating. Prime Minitrack observations were possible only during transmitter operation. Data included here are comprised of Prime Minitrack and Baker-Nunn observations made concurrently.

*Quinn*

## CONTENTS

Abstract . . . . .	ii
INTRODUCTION . . . . .	1
DISCUSSION OF DATA . . . . .	1
CONCLUSION . . . . .	2
ACKNOWLEDGMENTS . . . . .	3
References . . . . .	3
Appendix A—List of Symbols . . . . .	41

# PRIME MINITRACK AND BAKER-NUNN ORBITS OF SATELLITE 1959 $\alpha_1$ (VANGUARD II)

by

Hans G. Hertz

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## INTRODUCTION

Satellite 1959 $\alpha_1$  (Vanguard II) was launched 17 February 1959. The transmitter was operating for 26 days through 15 March 1959. Therefore Prime Minitrack observations could be obtained during this period. It was suggested to the author that a comparison of the accuracies of Prime Minitrack and Baker-Nunn observations of a satellite would be interesting. In the present report the data necessary for such a study are presented. The observations used are those made in the period where both types of observations were possible. There were 244 Prime Minitrack and 187 Baker-Nunn observations available which were made during this 26-day period.

## DISCUSSION OF DATA

For each of the two types, twelve orbits were determined. Their epochs were at 2-day intervals from 19 February 1959 through 13 March 1959 inclusive. For each orbit, the values of the parameters  $S_1, S_2, \dots, S_6, S_{18}$  were determined by differential corrections. Here  $S_1, S_2, \dots, S_6$  are the values of the constant terms in the expressions for the osculating elements  $a, e, I, \Omega, \omega, M$  as given by Brouwer (Reference 1). The quantity  $S_{18}$  is the coefficient of the term  $S_{18} t^2, \tau$  in units of 100 hours from the epoch, added to Brouwer's expression for the mean anomaly. In each differential correction all observations within 72 hours of the epoch have been used provided they generated residuals,  $\cos \delta \Delta \alpha$  and  $\Delta \delta$ , not larger than 0.10.

The earth parameters used in these solutions are shown in Table 1. The parameters are the equatorial radius  $R$  of the earth and the constants  $k, k_2, A_{30}, k_4, A_{50}$ . The last four constants occur in the expression,

$$U = \frac{\mu}{r} \left[ 1 + \frac{k_2}{r^2} (1 - 3 \sin^2 \beta) + \frac{A_{30}}{r^3} \left( -\frac{3}{2} \sin \beta + \frac{5}{2} \sin^3 \beta \right) \right. \\ \left. + \frac{k_4}{r^4} \left( 1 - 10 \sin^2 \beta + \frac{35}{3} \sin^4 \beta \right) + \frac{A_{50}}{r^5} \left( \frac{15}{8} \sin \beta - \frac{35}{4} \sin^3 \beta + \frac{63}{8} \sin^5 \beta \right) \right]$$

used by Brouwer. The constant  $k$  is given by  $\mu = k^2$ .

Several iterations were made. The orbits finally adopted as those based on the Prime Minitrack observations are called briefly Prime Minitrack orbits (PM orbits) and received the numbers 603 through 614. The numbers for the Baker-Nunn orbits (BN orbits) are 631, 616 through 625, and 632. The relationship between these numbers and the epochs is shown in Table 2.

The resulting values for the parameters  $S_1, S_2, \dots, S_6, S_{18}$  are shown in Table 3. There are twelve pairs of lines, one pair each belonging to one of the twelve epochs. The first line in each pair gives the results for the PM orbit, the second for the BN orbit. The orbit numbers given serve to identify the epochs.

Table 4 gives the probable errors for the  $S_i$  obtained. They are arranged in eight pairs of columns. The first column of each pair belongs to PM orbits, the second to BN orbits. One line corresponds to a PM and the corresponding BN orbit for the same epoch. The numbers of this PM and BN orbit are given in the first two columns of the table.

The residuals for the PM observations are shown in Table 5 and those for the BN observations in Table 6. Except for the observations near the beginning and end of the 26-day period every PM or every BN observation appears in three orbits. The PM observations have received serial numbers starting with 1. The numbers of the BN observations are those assigned by the Smithsonian Astrophysical Observatory, without the designation of the year. The observations being precision-reduced observations, all numbers begin with a '7'.

Table 7 gives information as to the accuracy of the representation of the observations by the adopted parameters. The weights given are those found in an iterative process in such a way that they are consistent with the probable errors computed from the residuals. This table shows that the rejection limit of 0.10 referred to on page 1 was too large. Better results would be achieved if it were lowered or if the rejection limit were made dependent on the distribution of errors.

By making additional runs the condition that no observations with residuals of more than 3 times the probable error be included was approximately met.

Table 8 shows the differences  $\Delta S_i$  of the values of the  $S_i$  obtained for a PM orbit and the corresponding BN orbit belonging to the same epoch. The differences are given in the sense PM-BN.

The  $S_i$  are plotted versus the time in Figures 1a-g for the Prime Minitrack orbits and in Figures 2a-g for the Baker-Nunn orbits. If no drag or other non-gravitational forces were present  $S_1, S_2, S_3$  would be constant,  $S_4, S_5, S_6$  would be linear functions of the time, and  $S_{18}$  would be zero. The probable errors in Table 4 and Figures 1a-f, 2a-f, and the non-vanishing of  $S_{18}$  indicate that there are deviations from gravitational behavior.

## CONCLUSION

Additional aspects of this problem are of interest and will be investigated if sufficient time and resources are available. For instance, the dependence of the  $S_i$  on the time could be investigated.

An examination of the  $\Delta S_i$  as to significance could be made. It would also be interesting to examine the residuals of the Prime Minitrack observations with respect to the Baker-Nunn orbits and the residuals of the Baker-Nunn observations with respect to the Prime Minitrack orbits. Finally, the data provide information on the relative accuracy of the two types of observations.

## ACKNOWLEDGMENTS

The computations on which the results of this report are based were carried out with a Differential Correction Program System and some additional programs. The original package was based on the satellite theory by H.G.L. Krause (Reference 2) and was programmed by Miss Elise R. Fisher of the Theoretical Division. Gratitude is also expressed to Mr. Cahill of the same division. The IBM Corporation under Dr. K. Deahl was utilized to substitute Brouwer's theory. Additional work was carried out under the supervision of Mr. A. Shapiro of GSFC. I thank Messrs. H. Bremer and R. Bryant of the Theoretical Division, Messers. R. Danek and J. Weld of the Data Systems Division, and others for help and advice received. Acknowledgement is also due the Smithsonian Astrophysical Observatory for providing the observations prior to publication.

## REFERENCES

1. Brouwer, Dirk, "Solution of the Problem of Artificial Satellite Theory Without Drag," *Astronomical Journal*, Vol. 64, no. 378, 1959.
2. Krause, H. G. L., "Die säkularen und periodischen Störungen der Bahn eines künstlichen Erdsatelliten," *Proceedings of the 7th International Astronautical Congress*, (1956) p. 523.

Table 1

Earth Parameters Used in Solutions.

R	6.378165	megameters
k	4118.0870	degrees megameters <sup>3/2</sup> hour <sup>-1</sup>
k <sub>2</sub>	0.02201451	megameters <sup>2</sup>
A <sub>30</sub>	0.00059678	megameters <sup>3</sup>
k <sub>4</sub>	0.00111709	megameters <sup>4</sup>
A <sub>50</sub>	0.00000000	megameters <sup>5</sup>

Table 2

Prime Minitrack and Baker-Nunn Orbits.

Epoch 0 <sup>h</sup> AT	J.D.	PM Orbit	BN Orbit
1959 February 19	2436618.5	603	631
21	6620.5	604	616
23	6622.5	605	617
25	6624.5	606	618
27	6626.5	607	619
March 1	6628.5	608	620
3	6630.5	609	621
5	6632.5	610	622
7	6634.5	611	623
9	6636.5	612	624
11	6638.5	613	625
13	6640.5	614	632

Table 3. Parameters of the Prime Minitrack and Baker-Nunn Orbits.

Orbit	S <sub>1</sub> Megameters	S <sub>2</sub>	S <sub>3</sub>	S <sub>4</sub>	S <sub>5</sub>	S <sub>6</sub>	S <sub>18</sub>
PM 603	8.3221388	0.16576178	32.872299	177.88713	142.47921	76.79972	+0.300991
BN 631	8.3221335	0.16584522	32.874176	177.89304	142.48830	76.79066	+0.300444
PM 604	8.3220465	0.16577543	32.873182	170.89423	153.01337	34.92399	+0.291683
BN 616	8.3220482	0.16577952	32.873129	170.89609	153.00516	34.93060	+0.308462
PM 605	8.3219558	0.16576787	32.874733	163.89978	163.53023	353.19875	+0.291654
BN 617	8.3219551	0.16577466	32.872571	163.90085	163.52863	353.19961	+0.300984
PM 606	8.3218575	0.16574035	32.874485	156.90739	174.06878	311.58485	+0.342934
BN 618	8.3218515	0.16576405	32.874596	156.90589	174.04075	311.61272	+0.356665
PM 607	8.3217416	0.16572010	32.873771	149.91296	184.57046	270.16722	+0.392135
BN 619	8.3217394	0.16573206	32.874683	149.91135	184.56912	270.17781	+0.356052
PM 608	8.3216274	0.16569527	32.873218	142.92069	195.08509	228.91773	+0.346715
BN 620	8.3216305	0.16573413	32.874890	142.91888	195.08849	228.92077	+0.338364
PM 609	8.3215251	0.16570116	32.872239	135.92960	205.59692	187.83316	+0.313567
BN 621	8.3215263	0.16572163	32.875379	135.92625	205.60918	187.82031	+0.329167
PM 610	8.3214260	0.16568639	32.872083	128.93408	216.11123	146.89236	+0.329278
BN 622	8.3214295	0.16571999	32.876015	128.93162	216.13230	146.86993	+0.302446
PM 611	8.3213280	0.16566514	32.873338	121.93588	226.64131	106.08772	+0.311599
BN 623	8.3213248	0.16575409	32.876659	121.93894	226.63302	106.08707	+0.381831
PM 612	8.3212291	0.16565315	32.875728	114.92644	237.17748	65.42991	+0.318442
BN 624	8.3212210	0.16572557	32.877521	114.93593	237.13776	65.47480	+0.270705
PM 613	8.3211282	0.16567097	32.875879	107.92496	247.70289	24.92186	+0.347865
BN 625	8.3211260	0.16575377	32.879222	107.93073	247.69734	24.93162	+0.333624
PM 614	8.3210186	0.16567475	32.876008	100.92886	258.22770	344.56841	+0.366178
BN 632	8.3210180	0.16576193	32.879606	100.93520	258.21165	344.58977	+0.356728

Table 4

Probable Errors of the Parameters of the Prime Minitrack and Baker-Nunn Orbits.

PM	BN	Orbit	Probable Error of									
			S <sub>1</sub> · 10 <sup>7</sup>	S <sub>2</sub> · 10 <sup>8</sup>	S <sub>3</sub> · 10 <sup>6</sup>	S <sub>4</sub> · 10 <sup>5</sup>	S <sub>5</sub> · 10 <sup>5</sup>	S <sub>6</sub> · 10 <sup>5</sup>	S <sub>18</sub> · 10 <sup>6</sup>			
603	631		19	839	1911	33	322	25	779	576	669	714
604	616		5	564	787	152	195	70	282	123	282	108
605	617		4	551	625	194	152	81	367	123	3820	1156
606	618		5	517	467	219	123	88	344	113	276	97
607	619		5	601	886	142	192	64	326	234	260	315
608	620		4	785	758	127	172	70	311	200	227	251
609	621		5	542	441	788	197	99	294	322	203	396
610	622		5	636	757	191	180	170	314	619	240	711
611	623		4	567	1094	131	270	161	408	561	334	522
612	624		5	614	713	309	233	202	332	707	190	707
613	625		4	756	217	545	237	49	341	226	189	245
614	632		3	589	583	207	209	59	291	270	139	313

Table 5  
Residuals of Prime Minitrack Observations.

Orbit Obs.	$\cos \delta \Delta \alpha$ (unit 0 <sup>0</sup> 0001)									
	603	604	605	606	607	608	609	610	611	614
1										
2	446									
3	323									
4	249									
5	78									
6	69									
7	53									
8	264	207								
9	400	131								
10	305	125								
11	1878	1653								
12	629	526								
13	1050	1496								
14	141	314								
15	1485	1596								
16	616	427								
17	310	509								
18	82	205								
19	631	486								
20	73	21								
21	221	14								
22	78	132								
23	213	437								
24	1623	1718								
25	180	6								
26	456	264								
27	603	384	409							
28	15	135	- 122							
29	441	240	293							
30	266	281	- 277							
31	681	519	589							
32	417	210	242							
33	115	176	176							
34	316	117	132							
35	1215	1352	- 1275							
36	197	124	152							
37	311	169	241							
38	222	51	- 38							
39	289	97	131							
40	49	80	38							
41	273	101	117							
42	211	58	104							
43	2084	2106	- 2007							
44	32	104	- 48							
45		701	726	456						
46		405	430	167						
47		418	395	684						
48		0	23	296						
49		270	362	334						
50		89	91	357						

Orbit Obs.	$\Delta\delta$ (unit 0".0001)									
	603	604	605	606	607	608	609	610	611	612
1	+ 225									614
2	+ 105									613
3	- 23									
4	+ 695									
5	- 22									
6	+ 311	+ 272								
7	- 89									
8	- 532	- 458								
9	- 314	- 320								
10	- 592	- 580								
11	- 268	- 301								
12	+ 110	+ 198								
13	+ 16	- 143								
14	0	+ 56								
15	+ 497	+ 603								
16	+ 1064	+ 1110								
17	+ 1227	+ 1194								
18	- 1512	- 1600								
19	- 745	- 684								
20	- 237	- 310								
21	- 125	- 87								
22	- 48	- 102								
23	- 126	- 208								
24	+ 382	+ 442								
25	- 42	- 123								
26	+ 987	+ 1050								
27	+ 940	+ 921	+ 976							
28	- 228	- 228	- 335							
29	- 611	- 186	- 484							
30	- 114	- 186	- 257							
31	+ 224	+ 336	+ 403							
32	+ 148	+ 128	+ 232							
33	- 231	- 282	- 142							
34	+ 248	+ 182	+ 205							
35	+ 456	+ 579	+ 646							
36	+ 147	+ 87	+ 21							
37	- 652	- 579	- 513							
38	- 290	- 335	- 414							
39	- 545	- 534	- 482							
40	+ 240	+ 160	+ 93							
41	- 738	- 806	- 792							
42	- 118	- 53	+ 9							
43	+ 504	+ 608	+ 682							
44	+ 901	+ 944	+ 1006							
45		+ 1490	+ 1525	+ 1436						
46		- 1729	- 1686	- 1736						
47		- 1386	- 1414	- 1593						
48		+ 364	+ 320	+ 147						
49		- 724	- 659	- 701						
50		+ 40	- 42	- 101						

Orbit	603	604	605	606	607	608	609	610	611	612	613	614
Obs.												

Orbit Obs.	603	604	605	606	607	608	609	610	611	612	613	614
51												
52		+ 276	+ 343	+ 190								
53		+ 180	+ 193	+ 117								
54		+ 197	+ 159	+ 88								
55		- 288	- 214	- 448								
56		+ 202	+ 269	+ 214								
57		+ 100	+ 148	+ 132								
58		- 1555	- 1560	- 1631								
59		- 835	- 514	- 491								
60		+ 258	+ 204	+ 152								
61		- 575	- 530	- 531								
62		- 26	- 90	- 130								
63		+ 117	+ 63	- 35								
64		- 14	- 101	- 79								
65		+ 543	+ 630	+ 17								
66		+ 118	+ 1073	+ 886								
67		+ 1006	+ 1067	+ 1065								
68			- 48	- 87	- 82							
69			+ 215	+ 168	+ 201							
70			- 1395	- 1419	- 1431							
71			- 480	- 459	- 476							
72			- 480	- 531	- 493							
73			- 3	+ 7	+ 327							
74			- 285	+ 295	+ 329							
75			- 909	- 1020	- 1116							
76			+ 450	+ 413	+ 451							
77			- 1523	- 1647	- 1556							
78			- 405	- 520	- 440							
79			- 94	- 154	- 102							
80			- 342	- 493	- 584							
81			- 108	- 40	- 67							
82			- 411	+ 404	+ 427							
83			- 28	- 73	- 181							
84			+ 66	+ 60	- 6							
85			+ 612	+ 546	+ 566							
86				- 467	- 382	- 268						
87				- 1187	- 1082	- 965						
88				+ 4	+ 88	+ 175						
89				- 206	- 265	- 324						
90				- 523	- 497	- 441						
91				- 53	+ 4	+ 56						
92				- 15	+ 16	+ 38						
93				+ 146	+ 369	+ 234						
94				- 40	+ 3	+ 58						
95				+ 249	+ 286	+ 327						
96				+ 186	+ 179	+ 177						
97				- 532	- 600	- 600						
98				- 124	- 148	- 154						
99				- 119	- 117	- 14						
100				+ 114	+ 202	+ 123						

$\Delta s$  (unit 0<sup>o</sup>0001)

[illegible]

Orbit Obs.	603	604	605	606	607	608	609	610	611	612	613	614
101					27	103						
102				188	180	196						
103				258	417	384						
104				438	903	905	936					
105					502	487	445					
106					385	416	487					
107					1893	1891	1801					
108					311	314	413					
109					653	672	709					
110					318	345	405					
111					605	594	581					
112					72	64	1					
113					2	41	65					
114					65	41	14					
115					193	213	190					
116					627	613	652					
117					924	873	878					
118					15	82	100					
119					327	273	236					
120					448	498	552					
121					519	462	514					
122					358	315	274					
123					90	109	147					
124					258	252	259					
125						669	704	742				
126						493	474	451				
127						43	623	56				
128						475	653	471				
129						521	595	622				
130						180	148	153				
131						96	130	126				
132						123	186	258				
133						220	162	149				
134						963	919	917				
135						671	722	729				
136						569	554	586				
137						803	778	771				
138						86	125	131				
139						886	937	904				
140						136	90	87				
141						154	202	202				
142						366	453	397				
143						101	123	143				
144						364	379	286				
145						187	172	172				
146						62	211	170				
147						122	18	12				
148						743	731	735				
149							946	925	939			
150							713	742	699			

Orbit Obs.	603	604	605	606	607	608	609	610	611	612	613	614
						$\cos \delta \Delta \alpha$ (unit 0.0001)						
151						86	153	141				
152						63	44	376				
153						-	48	122				
154						175	178	58				
155						54	20	71				
156						125	70	67				
157						232	345	425				
158						244	154	40				
159						207	103	20				
160						128	172	121				
161						127	157	141				
162						85	115	41				
163						574	546	613				
164						1090	1107	1234				
165						-	907	1004				
166						664	736	876				
167						910	939	1051				
168						-	286	212		78		
169						-	282	270		353		
170						-	121	334		730		
171						-	722	814		868		
172						-	138	59		23		
173						-	107	8		55		
174						-	138	178		109		
175						-	705	455		20		
176						-	760	128		455		
177						-	104	151		247		
178						-	225	242		272		
179						-	193	228		411		
180						-	65	51		2		
181						-	194	244		254		
182						-	145	23		179		
183						-	2	357		536		
184						-	492	15		172		
185						-		491		552	429	
186						-		164		256	-	
187						-		200		472	-	
188						-		285		81	-	
189						-		141		36	-	
190						-		134		176	-	
191						-		33		106	-	
192						-		178		131	-	
193						-		652		742	559	
194						-		205		214	-	
195						-		41		42	143	14
196						-		86		86	134	64
197						-		207		207	382	-
198						-		26		26	44	44
199						-		29		29	96	14
200						-		387		387	234	338
						-		129		129	195	37

Orbit Obs.	603	604	605	606	607	608	609	610	611	612	613	614
						$\Delta\delta$ (unit 0.0001)						
151						- 324		301	- 378			
152						- 812		830	- 725			
153						+ 226		+ 235	+ 150			
154						+ 214		+ 221	+ 187			
155						- 201		- 165	- 105			
156						+ 251		+ 360	+ 447			
157						+ 82		+ 96	+ 84			
158						+ 291		+ 282	+ 352			
159						- 662		- 655	- 644			
160						+ 76		+ 70	- 40			
161						- 833		- 832	- 714			
162						- 278		- 288	- 392			
163						+ 146		+ 152	+ 65			
164						+ 64		+ 64	+ 8			
165						- 41		+ 37	+ 147			
166						+ 154		+ 207	+ 235			
167						+ 152		+ 124	+ 157			
168								- 8	- 4	168		
169								+ 217	+ 124	- 110		
170								- 1169	- 1065	- 596		
171								+ 482	+ 433	+ 329		
172								+ 178	+ 109	- 36		
173								+ 160	+ 153	+ 157		
174								+ 84	+ 147	+ 376		
175								- 366	- 319	+ 213		
176								+ 148	+ 168	+ 751		
177								- 92	- 139	- 347		
178								+ 505	+ 457	+ 330		
179								- 686	- 581	- 645		
180								+ 426	+ 404	+ 399		
181								+ 60	+ 68	+ 217		
182								- 206	- 228	+ 64		
183								- 89	- 4	- 164		
184								- 879	- 709	- 853		
185									- 2	- 216	355	
186									- 280	- 475	- 582	
187									- 657	- 637	- 846	
188									+ 359	+ 308	+ 317	
189									- 327	- 250	- 174	
190									- 64	- 147	- 318	
191									- 179	- 403	- 496	
192									- 746	- 632	- 661	
193									- 453	- 207	- 135	
194									- 334	+ 22	+ 65	
195										- 167	- 254	- 335
196										- 540	- 461	- 423
197										- 184	- 226	- 265
198										- 124	- 119	- 113
199										- 214	- 178	- 122
200										- 135	- 112	- 84



Orbit Obs.	603	604	605	606	607	608	609	610	611	612	613	614	.
						$\Delta\delta$ (unit 0 <sup>o</sup> 0001)							.
201										- 243	- 323	- 364	
202										+ 117	- 42	- 120	
203										- 582	- 452	- 394	
204										- 377	- 438	- 474	
205										- 2364	- 2469	- 2481	
206										+ 960	+ 958	+ 957	
207										- 87	- 97	+ 104	
208										- 253	- 210	- 194	
209										+ 43	- 168	- 185	
210										- 124	- 159	- 148	
211										+ 337	+ 158	+ 198	
212										+ 445	+ 196	+ 191	
213										-	- 31	- 69	
214										-	- 233	- 249	
215										-	- 632	- 581	
216										+ 194	+ 189	+ 189	
217										+ 640	+ 642	+ 642	
218										- 138	- 124	- 124	
219										- 69	- 65	- 65	
220										- 338	- 388	- 388	
221										- 62	- 102	- 102	
222										- 275	- 222	- 222	
223										- 95	- 121	- 121	
224										- 126	- 114	- 114	
225										- 222	- 217	- 217	
226										+ 298	+ 339	+ 339	
227										- 990	- 941	- 941	
228										+ 404	+ 462	+ 462	
229										- 339	- 318	- 318	
230										+ 259	+ 200	+ 200	
231										- 1085	- 1203	- 1203	
232										-	- 210	- 210	
233										-	- 220	- 220	
234										-	- 306	- 306	
235										-	- 603	- 603	
236										-	- 405	- 405	
237										+ 283	+ 283	+ 283	
238										-	- 277	- 277	
239										-	- 165	- 165	
240										+ 228	+ 228	+ 228	
241										-	- 23	- 23	
242										-	- 407	- 407	
243										-	- 558	- 558	
244										-	- 282	- 282	

Table 6. Residuals of Baker-Nunn Observations.

Orbit Obs.	631	616	617	618	619	620	621	622	623	624	625	632
						cos $\delta \Delta \alpha$ (unit 0 <sup>o</sup> 0001)						
71138	23	29										
71414	11	12										
71132	19	18										
70993	7	3										
70992	4	19										
70991	6	15										
70896	2	26										
70100	13	1	10									
70131	25	10	+									
70102	20	8	+									
70103	16	6	+									
70104	3	5	-									
70329	9	8	-									
70340	4	7	+									
70961	29	16	-									
70962	25	9	0									
70015	615	522	-									
70813	31	144	+									
70757	67	42	+									
70756	86	34	+									
70791	-	7	-	35								
70792	-	5	+	46								
70793	-	11	-	27								
70394	+	18	+	22								
70022	-	9	-	9								
70023	-	9	+	7								
70024	-	20	-	28								
71599	-	6	-	16								
70493	-	21	-	308	380							
70494	-	55	+	286	+	376						
70495	-	32	+	310	+	404						
70492	-	26	+	316	+	413						
70064	+	5	+	57	+	62						
70065	-	1	+	38	+	47						
71006	-	125	-	83	-	87						
70072	-	137	-	102	-	104						
71342	-	244	-	356	-	109						
70576	-	-	-	22	+	102						
70360	-	-	-	58	+	61			183			
71362	-	-	-	40	+	74			+	140		
70480	-	-	-	55	+	56			+	150		
70484	-	-	-	50	+	56			+	131		
71363	-	-	-	927	-	828			-	129		
70748	-	-	-	39	+	13			-	758		
70750	-	-	+	14	-	13			+	65		
70751	-	-	+	21	-	7			+	41		
70083	-	-	-	141	-	181			-	51		
70084	-	-	-	144	-	181			-	122		
71519	+	35	-	35	-	183			-	123		
71405	+	22	+	22	-	26			-	7		
						29						

Orbit Obs.	631	616	617	618	619	620 $\Delta\delta$ (unit 0 <sup>o</sup> 0001)	621	622	623	624	625	632
71338	+	6										
71414	+	1	27									
71132	-	4	34									
70993	-	4	40									
70992	+	1	44									
70991	+	4	46									
70896	-	7	59									
70100	+	1	16	-								
70101	+	3	13	27								
70102	-	1	6	22								
70103	+	6	3	6								
70104	-	7	5	3								
70329	-	9	6	2								
70340	+	1	5	9								
70961	+	2	9	15								
70962	+	5	13	19								
70015	+	218	211	208								
70813	+	136	118	90								
70757	+	5	90	82								
70756	+	4	78	73								
70791	+		42	0								
70792	+	+	47	5	49							
70793	+	+	40	3	54							
70394	-	-	1	16	45							
70022	+	+	16	2	61							
70023	+	+	0	14	46							
70024	-	25	38	88	62							
71599	-	5	7	58	88							
70493	+		221	252	256							
70494	+		242	245	248							
70495	+	+	252	248	250							
70492	+		265	253	254							
70064	+	+	81	35	13							
70365	+	+	87	36	15							
71006	+	+	32	54	66							
70072	+	+	54	37	47							
71342	-		10	26	60							
70576	-			20	27							
70360	-			3	11							
71362	-			3135	3144							
70480	-			2	12							
70484	-			1	13							
71363	-			381	395							
70748	+	+	+	2	29							
70750	+	+	8	8	24							
70751	+	+	6	6	25							
70083	+	+	29	29	4							
70084	+	+	9	9	19							
71519	+	+	51	51	12							
71405	+	+	46	46	17							

Orbit	631	616	617	618	619	$\cos \delta \Delta \alpha$	620	621	622	623	624	625	632
Obs.								(unit 0.0001)					
71272				35	-	-	49						
70466			+	34	-	-	50						
70467			+	28	-	-	56						
70086			+	24	-	-	59						
70087			+	80	+	+	24						
71189			+	75	+	+	17						
71596			+	24	-	-	29						
71597			-	10	-	-	47						
71155			-	78	-	-	141						
70037			-	96	-	-	176						
71156			-	99	-	-	179						
70038			-	92	-	-	172						
72030			-		+	+	8	40					
72031			+		+	+	4	43					
70482			+		+	+	7	48					
70438			+		+	+	2	36					
70396			+		+	+	36	37					
70397			+		+	+	41	42					
70398			+		+	+	8	9					
70868			+		+	+	20	21					
71247			+		+	+	9	11					
71246			+		+	+	4	6					
71642			+		+	+	4	1					
71986			+		+	+	17	42					
70970			-		+	+	44	64					
71205			+		+	+	11	3					
71331			+		+	+	8	7					
71206			+		+	+	0	16					
70092			+		+	+	14	36					
70975			+		+	+	17	42					
70091			-		+	+	134	163					
71195			-		+	+	133	162					
70485			-		+	+	64	42					
70787			-		+	+	114	97					
71364			-		+	+	560	545					
70341			-		+	+	66	55					
70651			+		+	+	11	82					
70039			+		+	+	62	14					
71313			+		+	+	60	36					
71152			-		+	+	41	33					
71314			-		+	+	108	122					
70344			-		+	+	28	33					
70111			+		+	+	93	7	1				
70112			+		+	+	94	6	1				
71473			+		+	+	95	8	3				
71520			+		+	+	131	53	60				
70095			+		+	+	103	21	28				
71516			+		+	+	147	57	64				
70113			+		+	+	234	128	130				
70964			+		+	+	205	99	101				

Orbit	617	618	619	$\Delta\delta$ (unit 0 <sup>o</sup> 0001)	620	621	622	623	624	625	631
Obs.											
71272	+	28	-	6	-	7					
70466	+	28	-	5	-	7					
70467	+	30	-	0	-	2					
70086	+	25	-	5	-	6					
70087	+	51	-	18	+	13					
71189	+	57	+	25	+	19					
71196	+	57	+	57	+	47					
71597	+	67	+	73	+	63					
71155	-	6	-	38	-	50					
70037	-	41	-	66	-	78					
71156	-	1	-	21	-	33					
70038	+	6	-	9	-	19					
72030	+		+	15	-	6					
72031	+		+	47	+	26					
70482	+		+	43	+	21					
70438	+		+	24	+	2					
70396	-		-	60	-	66					
70397	-		-	58	-	64					
70398	-		-	57	-	62					
70868	-		-	58	-	63					
71247	-		-	54	-	58					
71246	-		-	59	-	62					
71642	-		-	68	-	71					
71986	+		+	12	+	7					
70970	-		-	5	-	5					
71205	+		+	36	+	20					
71331	+		+	30	+	14					
71206	+		+	35	+	18					
70092	+		+	38	+	23					
70975	+		+	38	+	28					
70091	+		+	21	+	2					
71195	+		+	27	+	9					
70485	+		+	50	+	29					
70787	-		+	27	+	5					
71364	+		+	140	-	162					
70341	+		+	40	+	17					
70651	+		+	48	+	39					
70039	+		+	49	+	42					
71313	-		-	76	-	68					
71152	-		-	76	-	72					
71314	-		-	99	-	102					
70344	-		-	26	-	37					
70111	+		+		+	44					
70112	+		+		+	44					
71473	+		+		+	45					
71520	+		+		+	67					
70095	+		+		+	63					
71516	+		+		+	41					
70113	+		+		+	37					
70964	+		+		+	26					

Orbit Obs.	631	616	617	618	619	620	621	622	623	624	625	632
						$\cos \delta \Delta \alpha$	(unit 0.0001)					
76905						+ 180	+ 88	+ 52				
76779						+ 327	+ 244	+ 170				
71085						+ 345	+ 259	+ 183				
70115						+ 377	+ 290	+ 214				
71521						+ 357	+ 269	+ 194				
70117						+ 348	+ 260	+ 185				
70957						+ 54	+ 94	+ 53				
70815						- 349	- 408	- 483				
71008						- 332	- 392	- 467				
70120						- 287	- 347	- 420				
70816						- 237	- 294	- 362				
70121						- 184	- 239	- 304				
71655						- 284	- 325	- 383				
70044						- 342	- 388	- 450				
71653						- 382	- 436	- 506				
72123						- 396	- 457	- 534				
70045						- 180	- 180	- 229				
70123						- 137	+ 5	- 50	328			
74186							+ 74	+ 18	- 268			
71028							+ 75	+ 19	- 272			
70837							+ 73	+ 16	- 281			
76907							+ 172	+ 144	- 52			
70125							- 438	- 447	- 568			
70126							- 294	- 299	- 484			
71647							- 219	- 220	- 406			
70127							- 149	- 147	- 328			
71593							+ 142	+ 206	+ 201			
71474							+ 151	+ 216	+ 204			
71318							+ 152	+ 217	+ 198			
71218							+ 153	+ 221	+ 184			
71022							+ 168	+ 235	+ 196			
71021							+ 157	+ 224	+ 183			
76908							+ 98	+ 180	+ 224			
76909							+ 131	+ 214	+ 255			
76910							- 1022	- 940	- 902			
70128							- 362	- 183	- 269			
71643								- 101	- 116	54		
70131								+ 1101	+ 1077	+ 1236		
71648								- 120	- 174	- 46		
71649								- 50	- 151	- 96		
70051									+ 116	+ 73	7	
70744									+ 132	+ 86	+ 9	
70555									+ 129	+ 64	+ 6	
71594									+ 126	+ 59	+ 4	
70052									+ 123	+ 28	+ 11	
70132									+ 128	0	+ 115	
71027									+ 70	+ 111	+ 125	
70134									- 116	+ 64	+ 82	52
70141									- 80	+ 35	- 9	-
70144									- 56	-	-	10

21



Orbit Obs.	631	616	617	618	619	620	621	622	623	624	625	632
						$\Delta\delta$ (unit 0.00001)						
70146										+ 30	+ 20	+ 10
70147										- 4	- 18	- 26
70349										+ 109	+ 109	+ 101
71675										+ 79	+ 67	+ 65
70151										- 19	+ 15	+ 13
70976										- 2	+ 27	+ 27
70977										+ 9	+ 30	+ 33
72025											+ 16	- 10
71639											+ 12	- 14
71616											- 118	- 137
72024										+ 20	+ 20	- 4
71400										- 18	- 18	- 30
71040										- 16	- 16	- 29
71384										0	0	- 14
71037										- 10	- 10	- 24
71038										- 8	- 8	- 22
70817										- 15	- 15	- 20
71674										- 19	- 19	- 23
71266										- 14	- 14	- 16
72469										+ 4	+ 4	- 12
72466										+ 17	+ 17	0
72039										+ 15	+ 15	- 2
72625										+ 15	+ 15	- 2
72040										+ 15	+ 15	- 1
70332										- 170	- 170	- 158
71262												- 172
70056											+ 59	+ 59
70158											+ 39	+ 39
70843											+ 33	+ 33
70162											+ 44	+ 44
70164											+ 39	+ 39
70844											+ 574	+ 574
70745											+35403	+35403
70468											+34673	+34673
70165											+33529	+33529
70166											+ 22	+ 22
70167											+ 17	+ 17

Table 7. Accuracy of Representation of Observations

		Orbit											
		603	604	605	606	607	608	609	610	611	612	613	614
PM	BN	631	616	617	618	619	620	621	622	623	624	625	632
Probable Error of Observation of Weight 1													
PM	BN	0.0100	0.0103	0.0104	0.0099	0.0108	0.0124	0.0115	0.0116	0.0111	0.0110	0.0128	0.0104
		0.0004	0.0013	0.0009	0.0031	0.0033	0.0035	0.0045	0.0041	0.0022	0.0021	0.0016	0.0022
Weight in R.A.													
PM	BN	1.45	1.44	1.18	0.57	1.51	1.29	1.50	1.33	1.64	1.59	1.38	1.63
		0.06	1.65	1.80	0.86	0.55	0.23	0.32	0.12	0.03	0.20	0.31	0.19
Weight in Decl.													
PM	BN	0.55	0.56	0.82	1.43	0.49	0.71	0.50	0.67	0.36	0.41	0.62	0.37
		1.94	0.35	0.20	1.14	1.45	1.77	1.68	1.88	1.97	1.80	1.69	1.81
Probable Error of Observation in R.A.													
PM	BN	0.0083	0.0086	0.0095	0.0130	0.0088	0.0109	0.0094	0.0101	0.0087	0.0087	0.0109	0.0081
		0.0018	0.0010	0.0007	0.0033	0.0044	0.0073	0.0080	0.0118	0.0140	0.0047	0.0028	0.0051
Probable Error of Observation in Decl.													
PM	BN	0.0135	0.0137	0.0115	0.0083	0.0154	0.0147	0.0163	0.0142	0.0185	0.0172	0.0163	0.0170
		0.0003	0.0022	0.0021	0.0029	0.0027	0.0026	0.0035	0.0030	0.0016	0.0016	0.0012	0.0017

Table 8. Differences of Parameters

PM Orbit	$\Delta S_1$ Megameters	$\Delta S_2$	$\Delta S_3$	PM-BN $\Delta S_4$	$\Delta S_5$	$\Delta S_6$	$\Delta S_{18}$
603	+0.0000053	-0.000008344	- 0°001877	- 0°00591	- 0°00909	+ 0°00906	+0°000547
604	-0.0000017	-0.00000409	+ 0°000053	- 0°00186	+ 0°00821	- 0°00661	-0°016779
605	+0.0000007	-0.00000679	+ 0°002162	- 0°00107	+ 0°00160	- 0°00086	-0°009330
606	+0.0000060	-0.00002371	- 0°000111	+ 0°00150	+ 0°02803	- 0°02787	-0°013731
607	+0.0000022	-0.00001196	- 0°000912	+ 0°00161	+ 0°00134	- 0°01059	+0°036083
608	-0.0000031	-0.00003885	- 0°001672	+ 0°00181	- 0°00340	- 0°00304	+0°008351
609	-0.0000012	-0.00002047	- 0°003140	+ 0°00334	- 0°01226	+ 0°01285	-0°015600
610	-0.0000035	-0.00003360	- 0°003932	+ 0°00246	- 0°02107	+ 0°02243	+0°026832
611	+0.0000032	-0.00000895	- 0°003321	- 0°00306	+ 0°00829	+ 0°00065	-0°070232
612	+0.0000081	-0.00007242	- 0°001793	- 0°00949	+ 0°03972	- 0°04489	+0°047737
613	+0.0000022	-0.00000280	- 0°003343	- 0°00577	+ 0°00555	- 0°00976	+0°014241
614	+0.0000006	-0.000008718	- 0°003598	- 0°00634	+ 0°01605	- 0°02136	+0°009450

Figure 1a through 1g— $S_1$  versus time for Prime Minitrack Orbits.

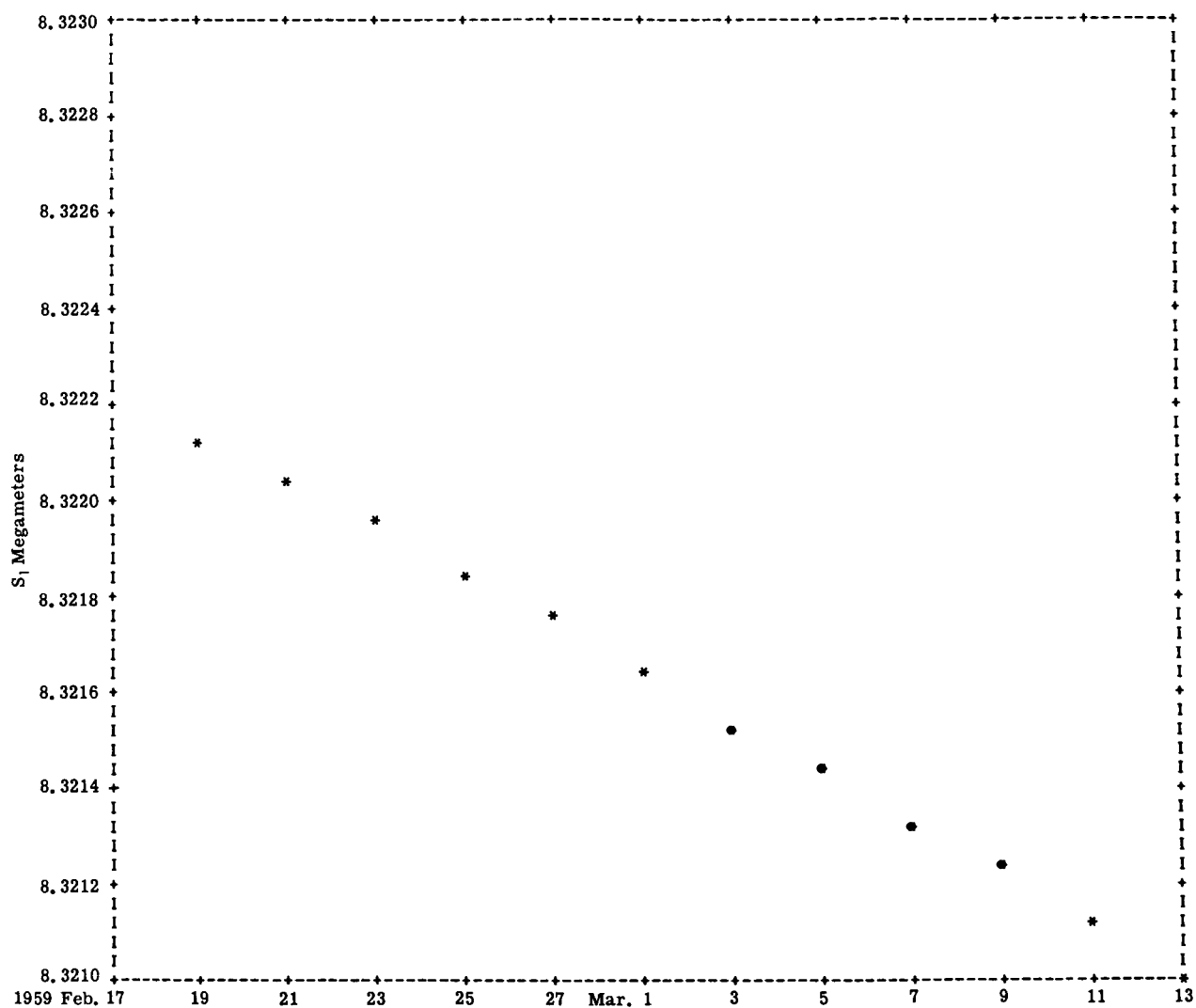


Figure 1a—Values of  $S_1$  (megameters)

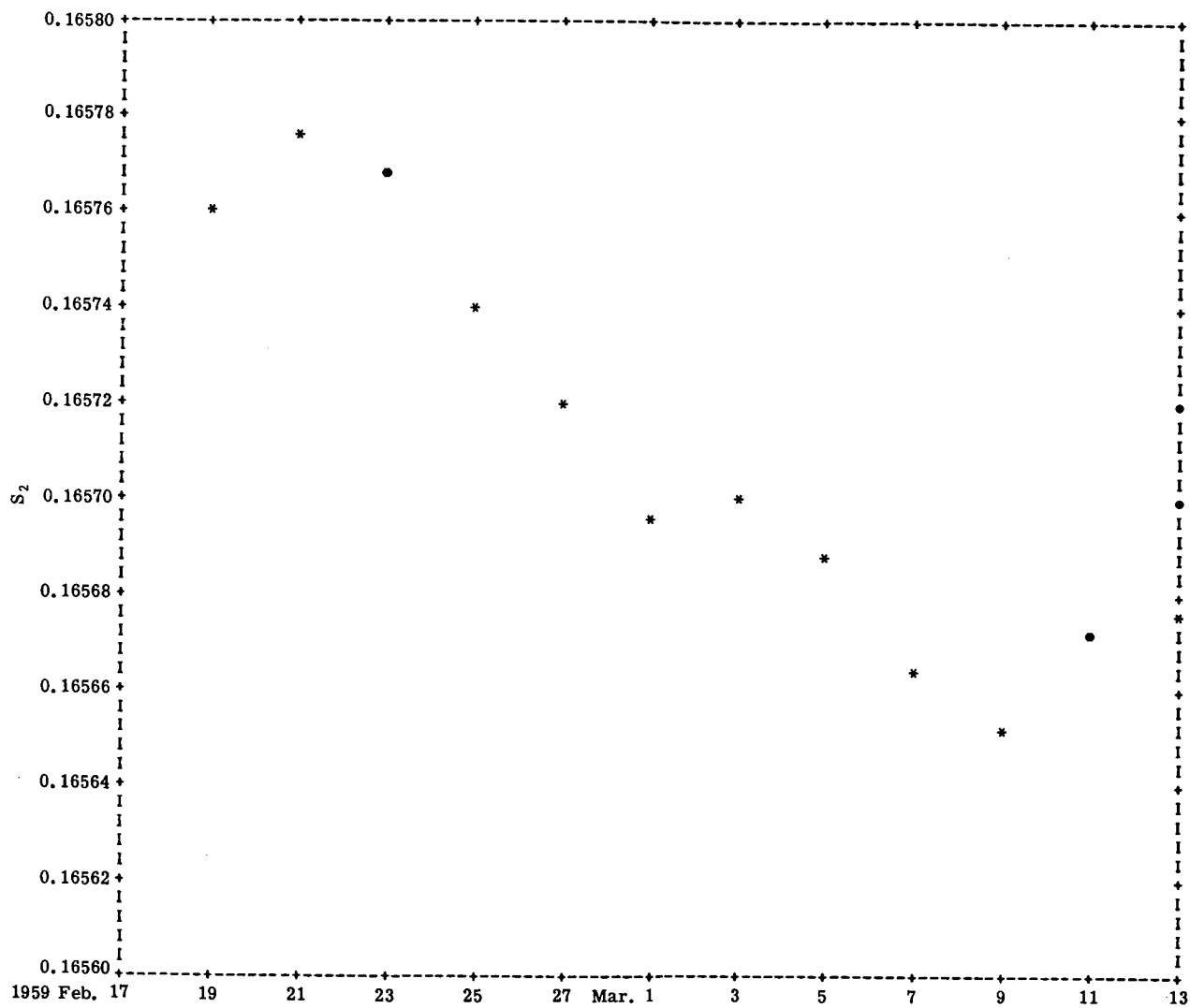


Figure 1b—Values of  $S_2$

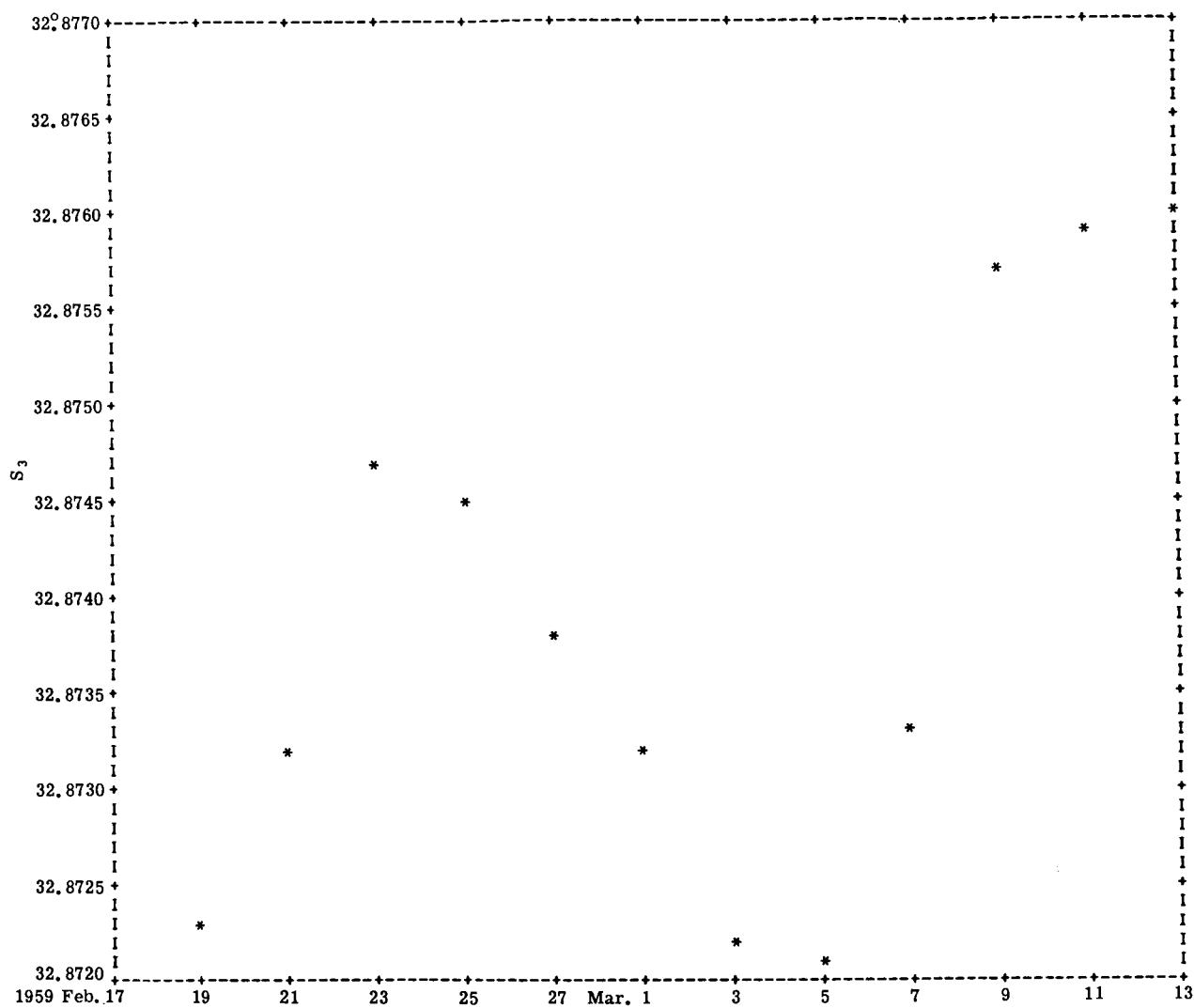


Figure 1c—Values of  $S_3$  (degrees)

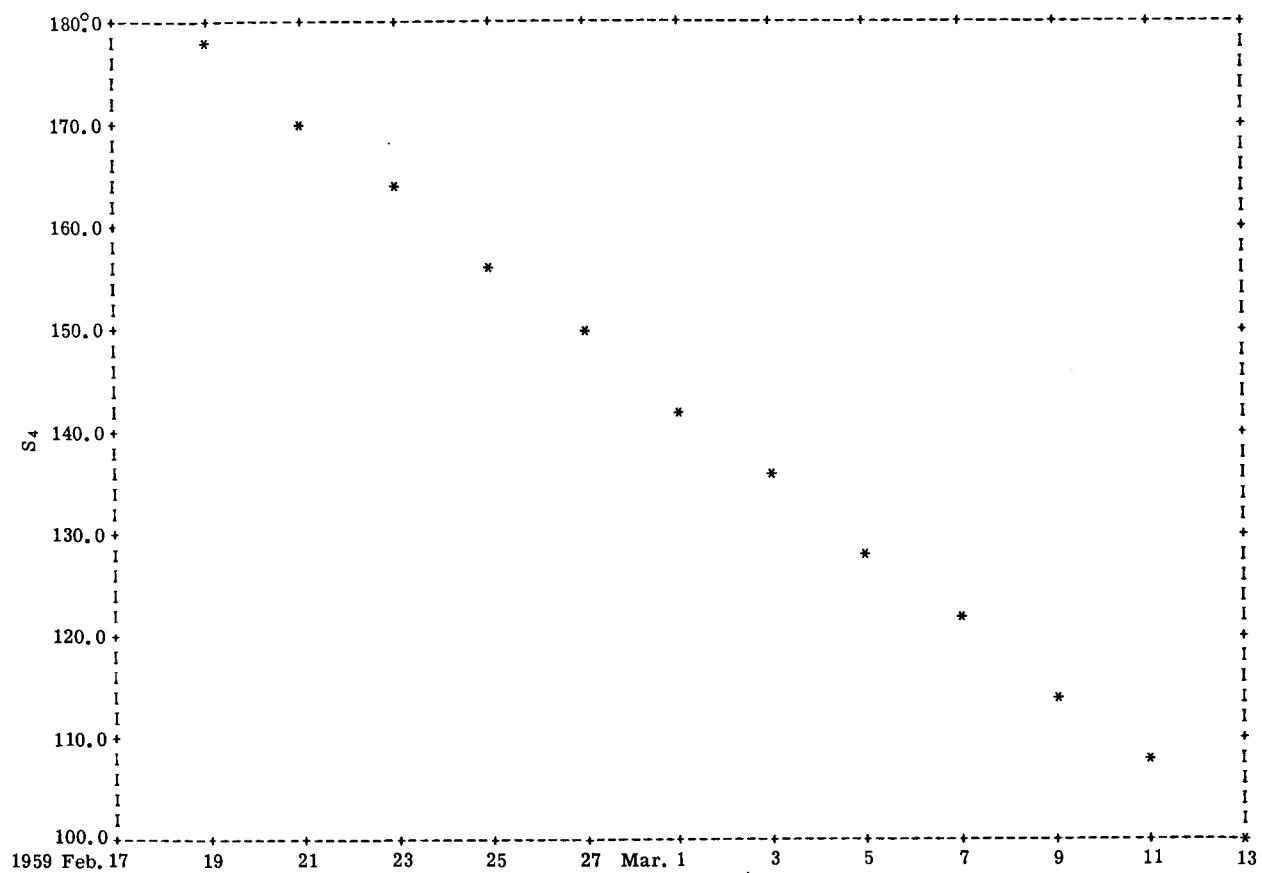


Figure 1d—Values of  $S_4$  (degrees)

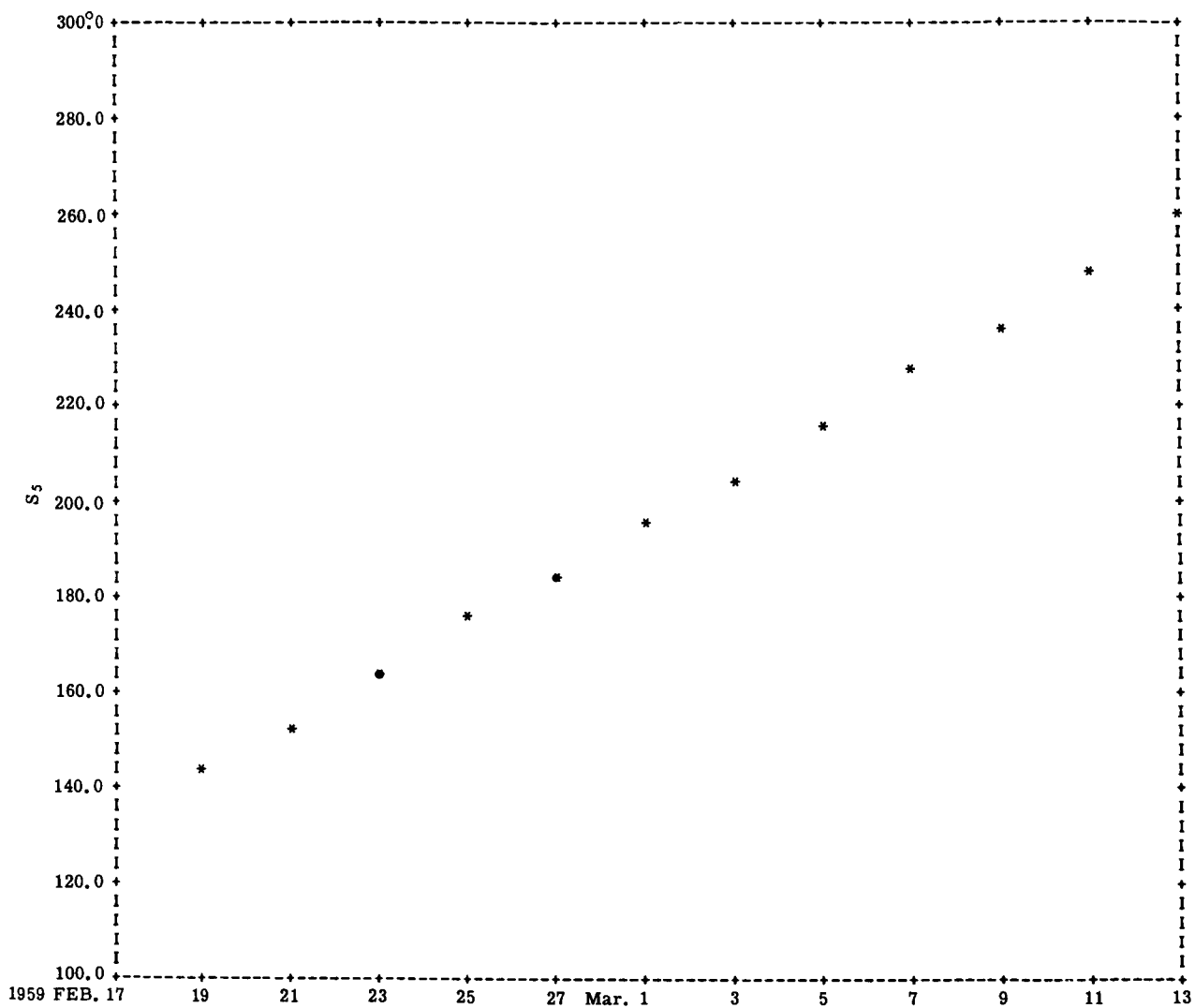


Figure 1e—Values of  $S_s$  (degrees)

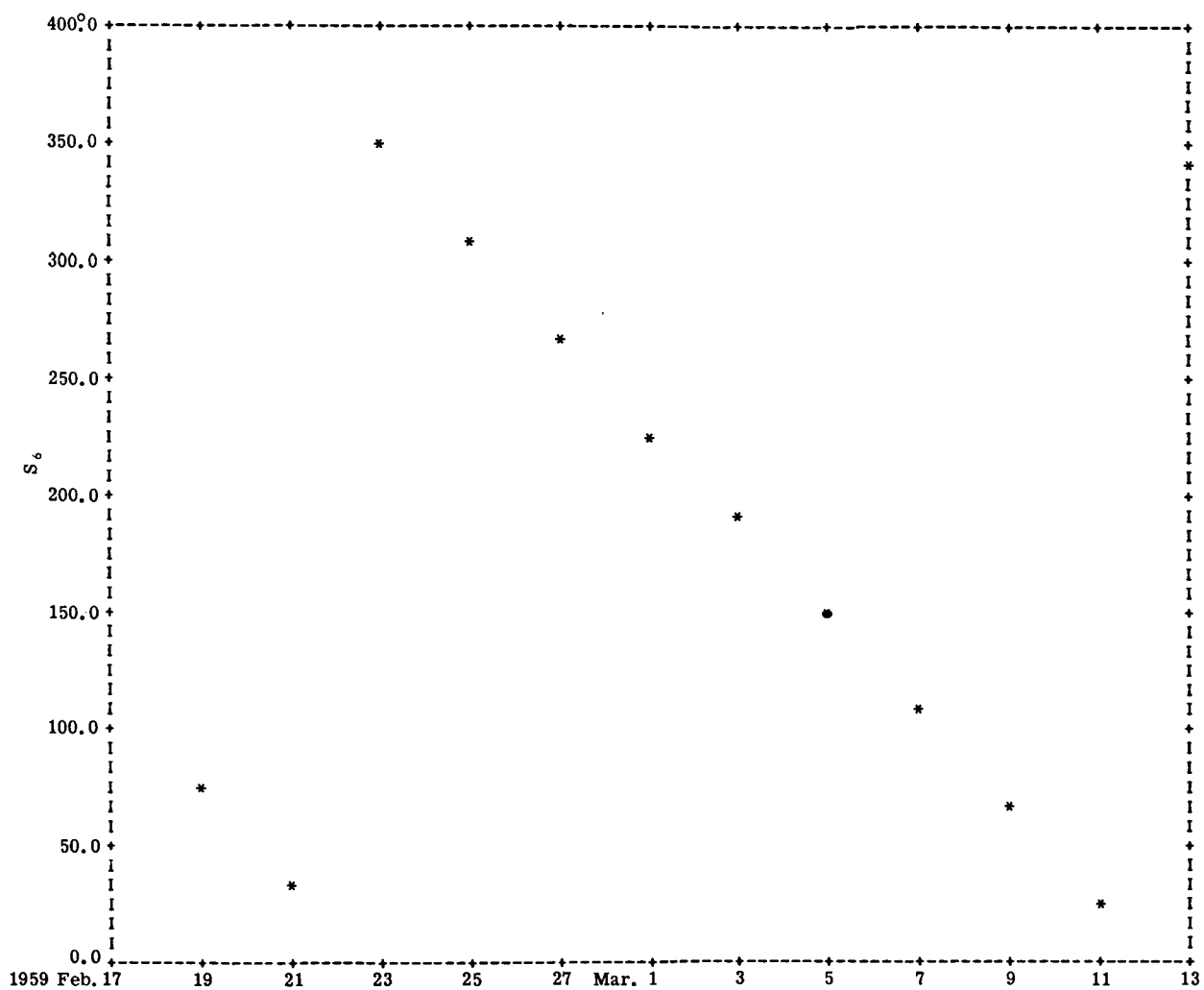


Figure 1f—Values of  $S_\delta$  (degrees)

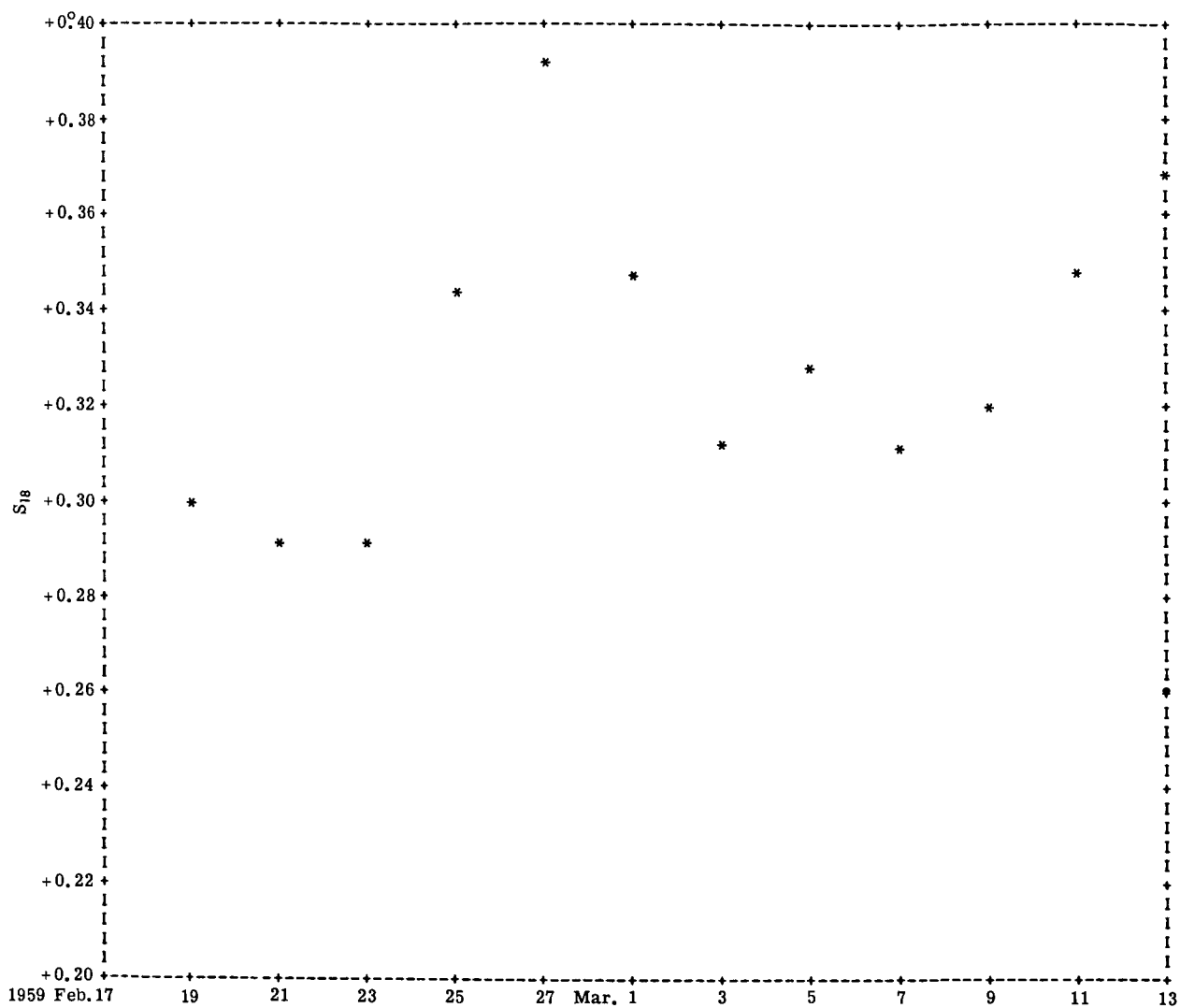


Figure 1g—Values of  $S_{18}$  (degrees)

Figure 2a through 2g— $S_i$  versus time for Baker-Nunn Orbits.

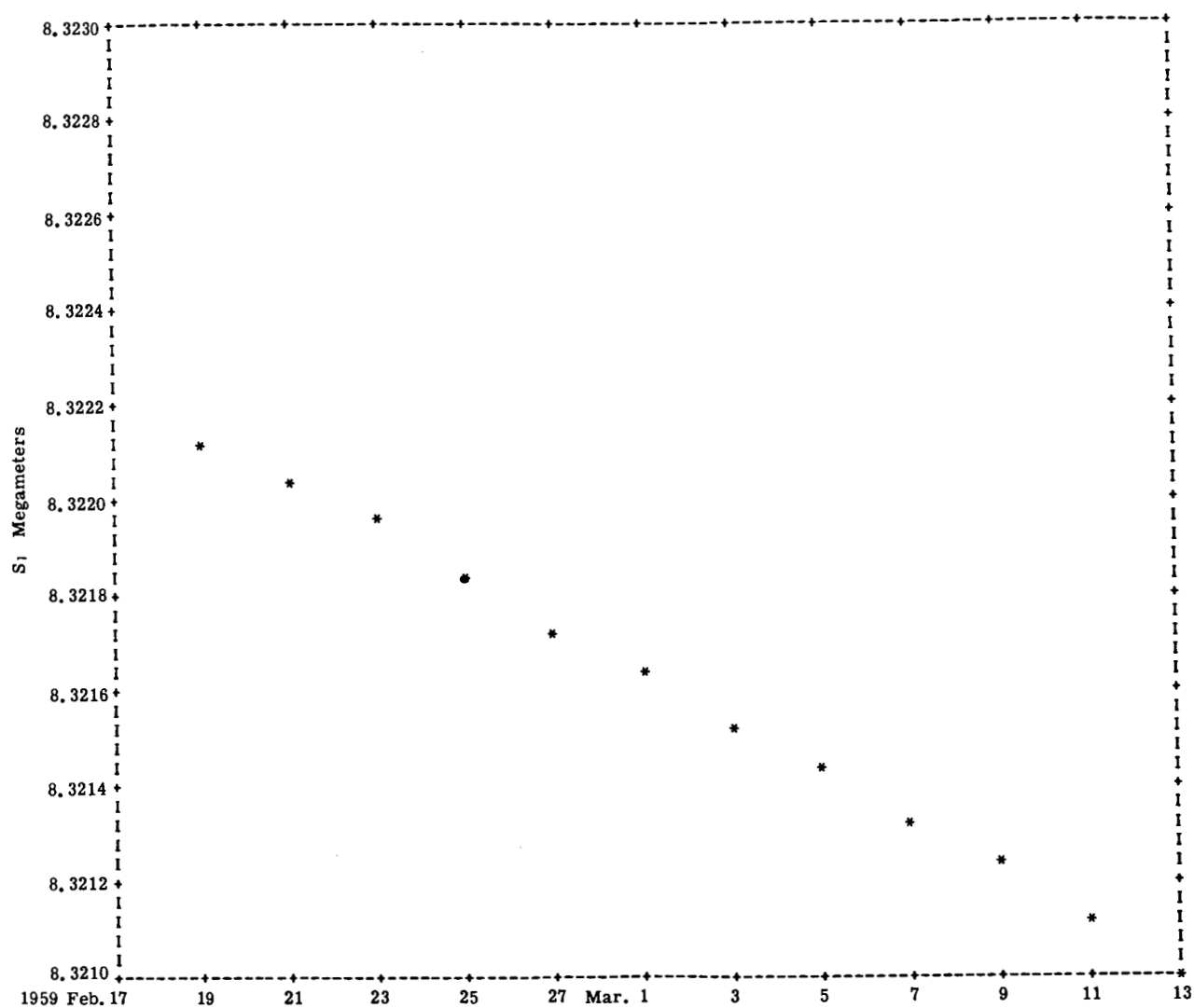


Figure 2a—Values of  $S_1$  (megameters)

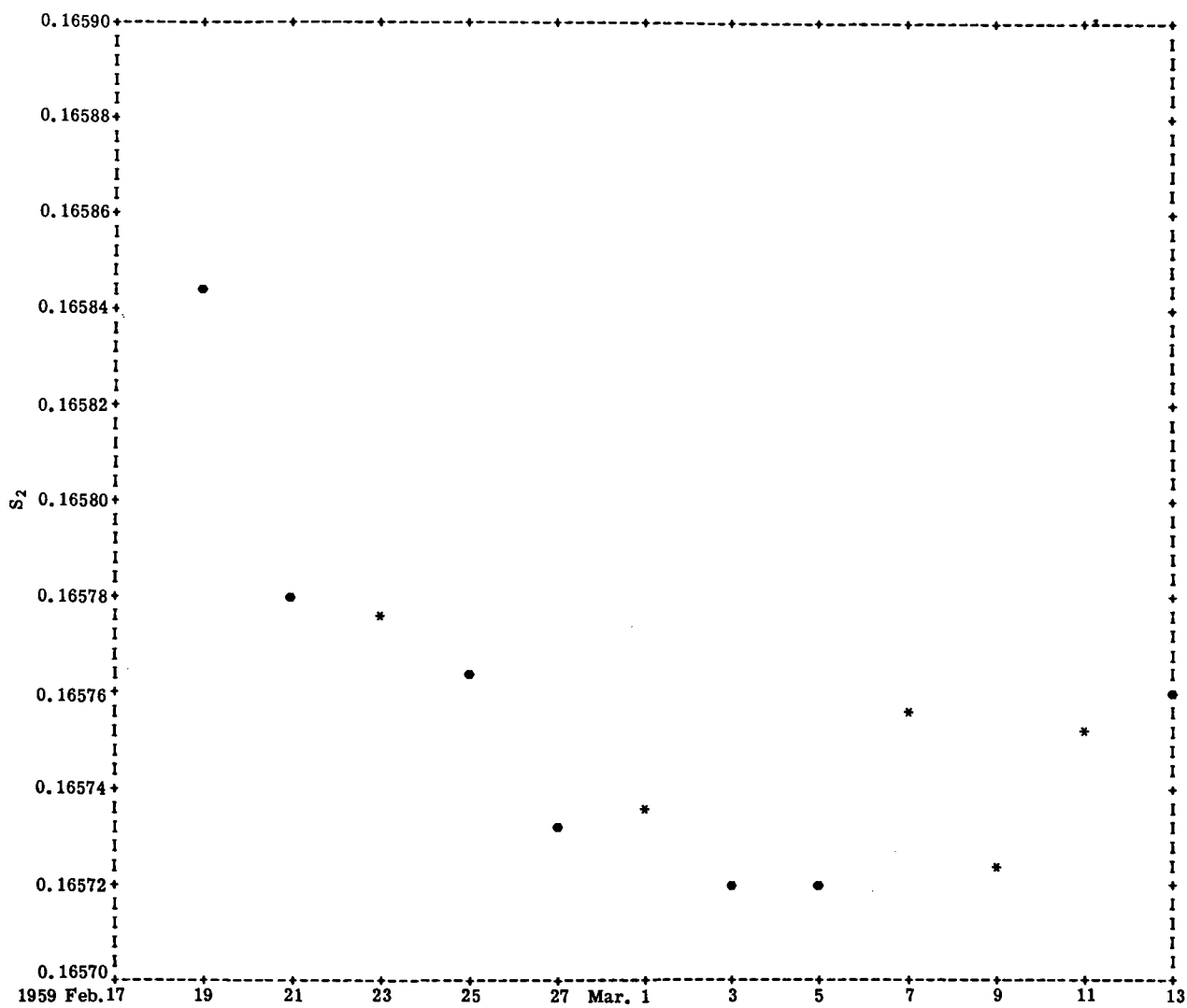


Figure 2b—Values of  $S_2$

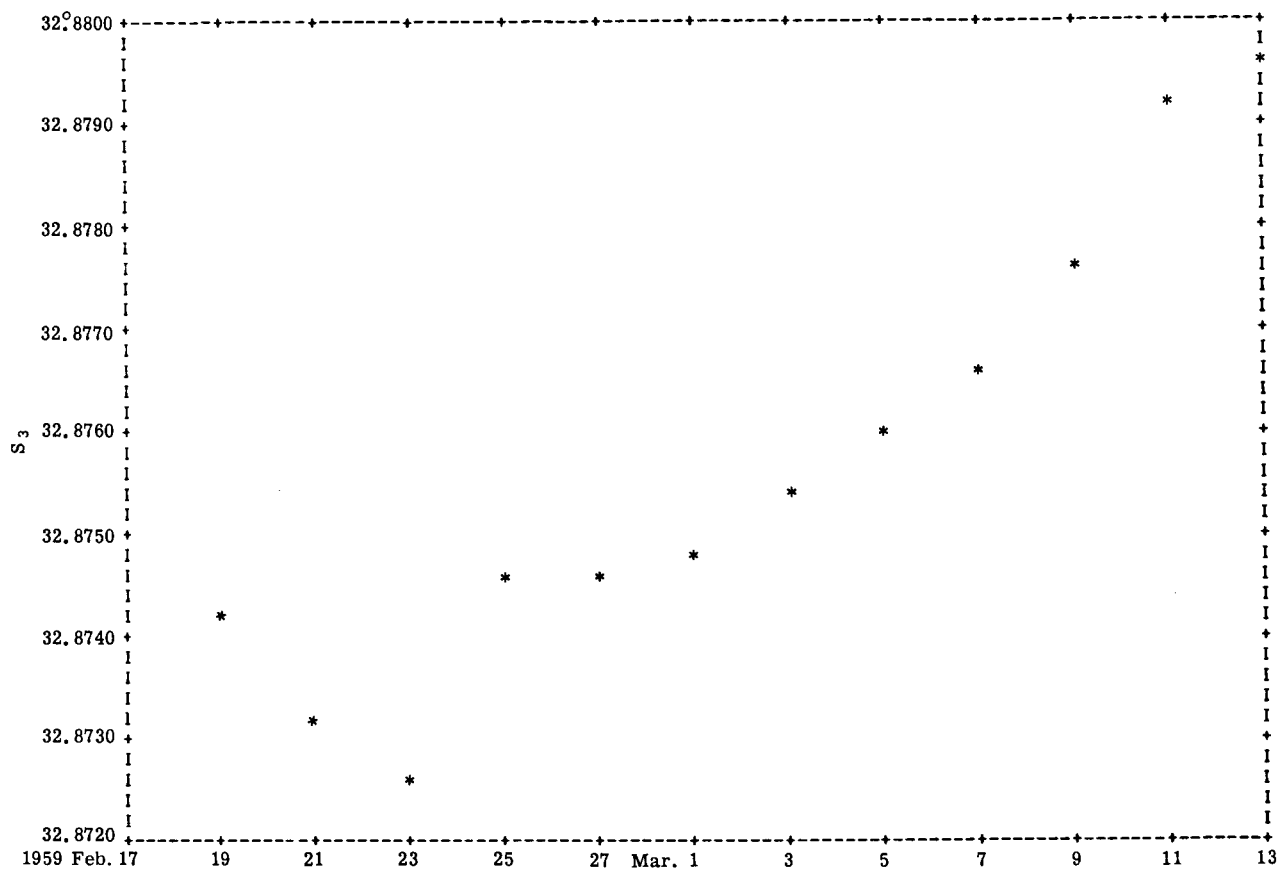


Figure 2c—Values of  $S_3$  (degrees)

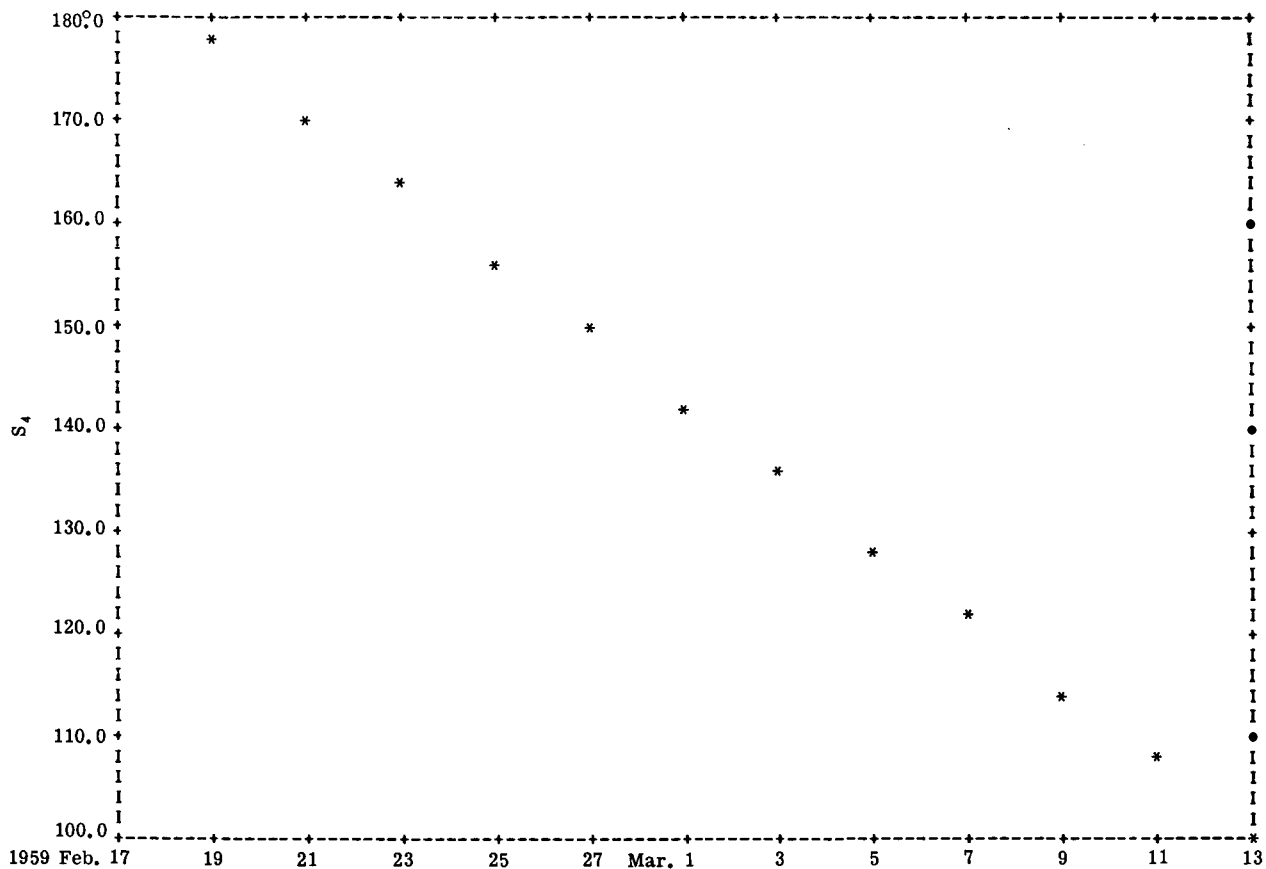


Figure 2d—Values of  $S_4$  (degrees)

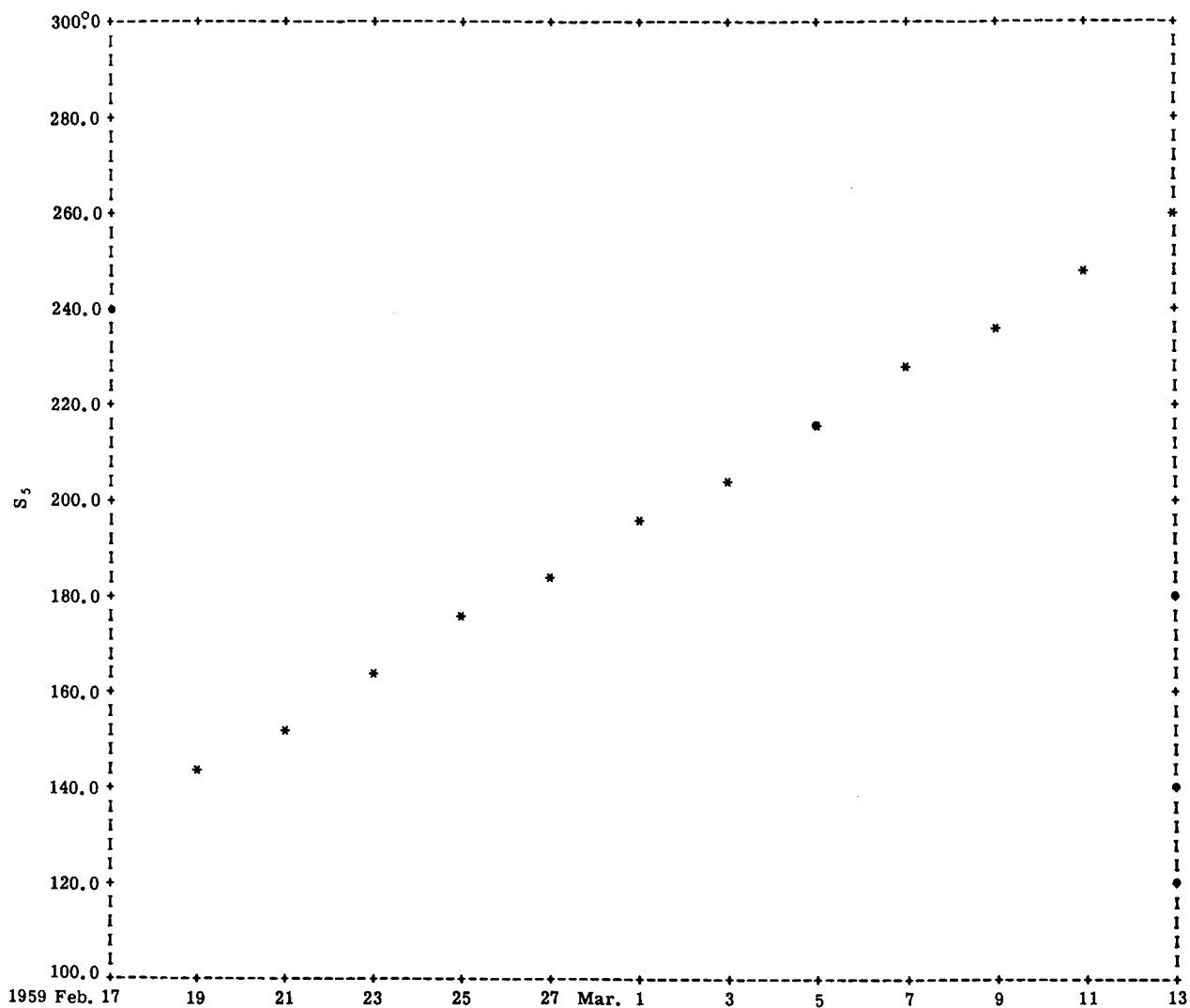


Figure 2e—Values of  $S_5$  (degrees)

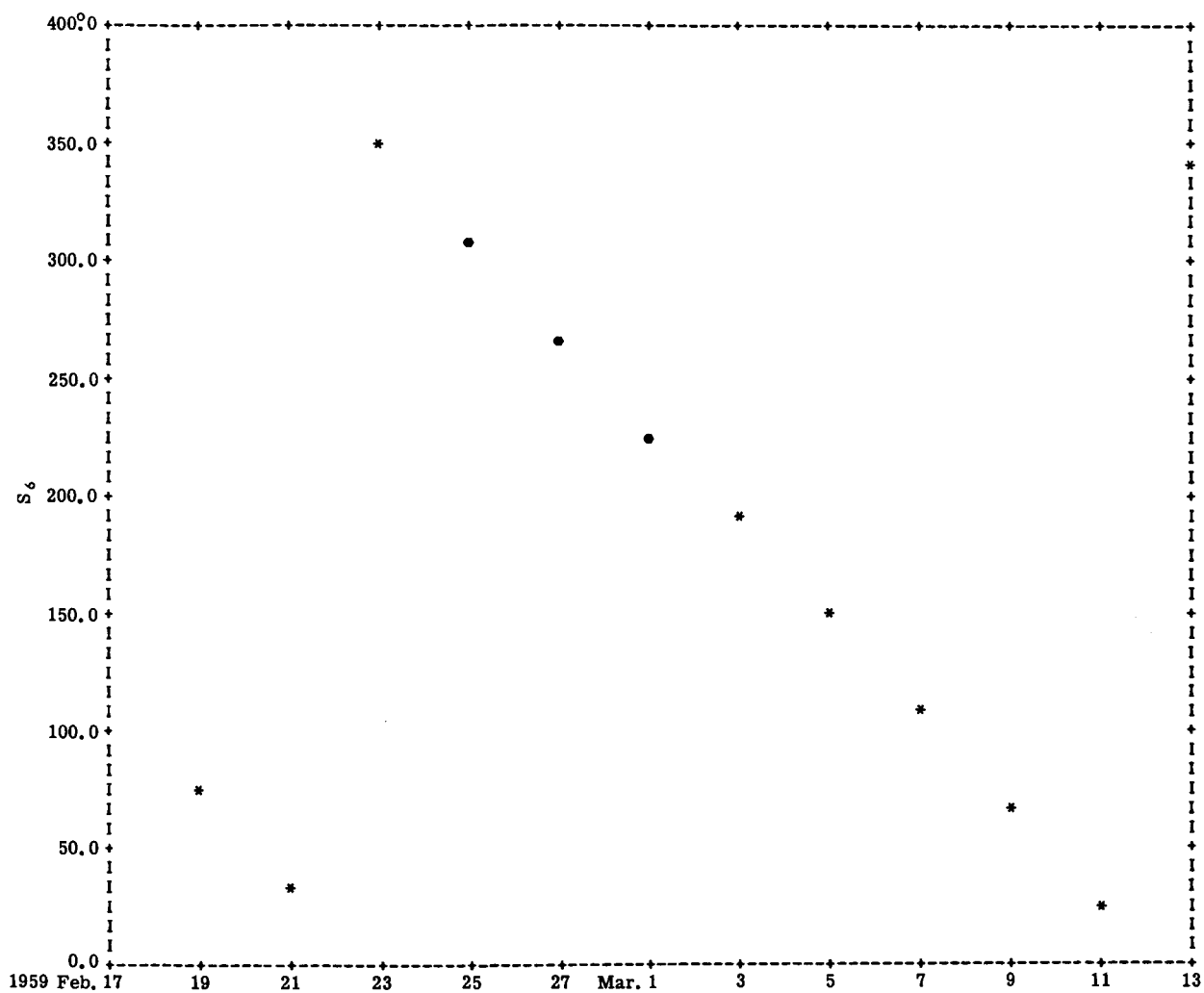


Figure 2f—Values of  $S_6$  (degrees)

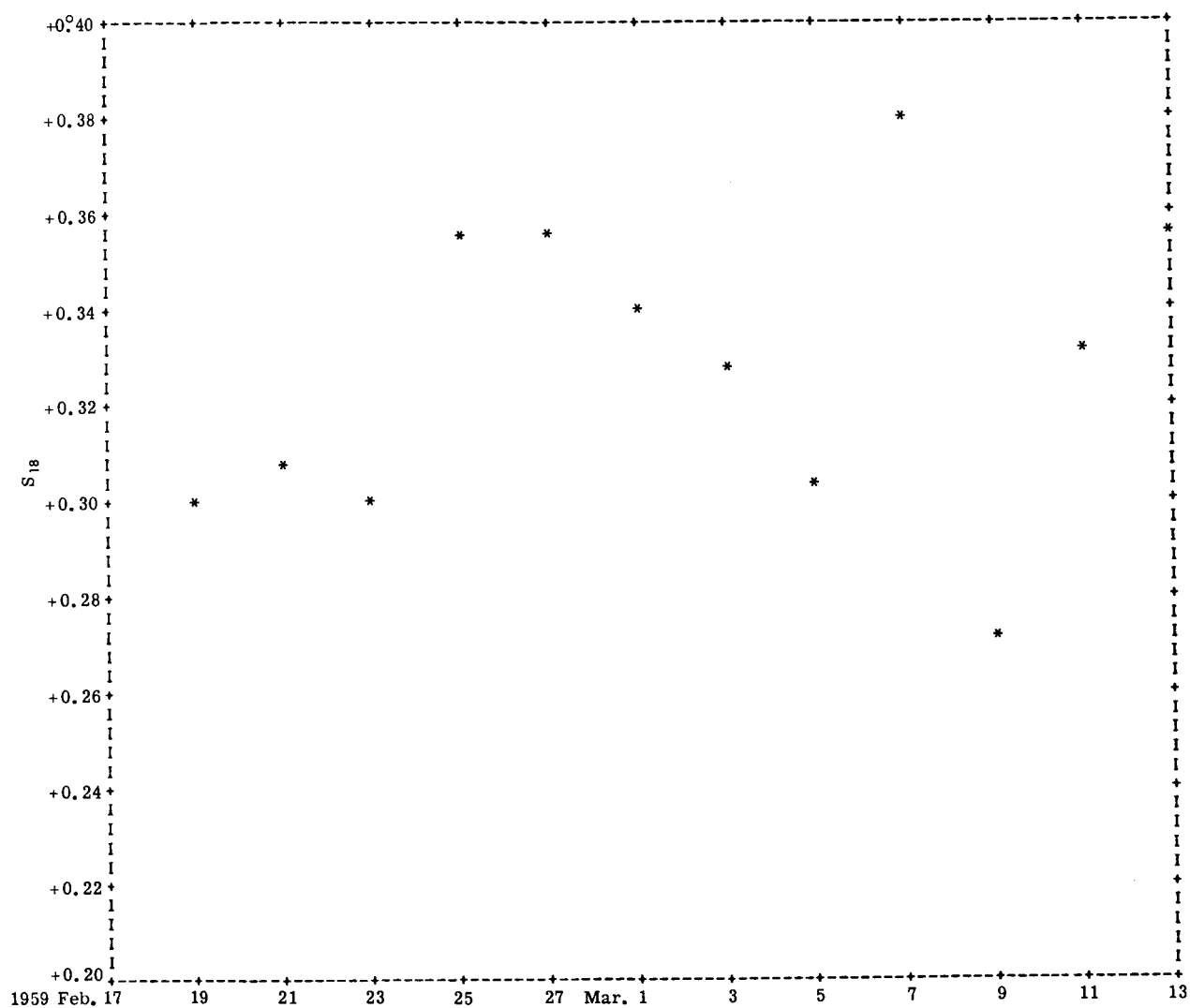


Figure 2g—Values of  $S_{18}$  (degrees)

Appendix A  
List of Symbols

Symbol	Meaning
$A_{30}$	Constant for earth's potential (megameters <sup>3</sup> )
$A_{50}$	Constant for earth's potential (megameters <sup>5</sup> )
$I$	Inclination of orbital plane to equator (degrees)
$M$	Mean anomaly
$R$	Earth equatorial radius (megameters)
$a$	Semimajor axis (megameters)
$e$	Eccentricity (non-dimensional)
$k$	Gravitational constant (degrees megameters <sup>3/2</sup> hour <sup>-1</sup> )
$k_2$	Constant for earth's potential (megameters <sup>2</sup> )
$k_4$	Constant for earth's potential (megameters <sup>4</sup> )
$\Delta\delta$	Residual in declination
$\Omega$	Longitude of ascending node
$\alpha$	Right ascension
$\beta$	Declination of point for which potential is considered
$\delta$	Declination
$\mu$	$k^2$ (degrees <sup>2</sup> megameters <sup>3</sup> hours <sup>-2</sup> )
$\tau$	Time in units of one-hundred hours
$\omega$	Argument of perigee
$\cos \delta \Delta\alpha$	Residual in right ascension